

# AVIATION

*The Oldest American Aeronautical Magazine*

NOVEMBER 17, 1924

Issued Weekly

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VOLUME  
XVII

## SPECIAL FEATURES

NUMBER  
20

TWO-PLACE LIGHT PLANES IN ENGLAND  
DRIGGS-JOHNSON LIGHT PLANE DESCRIBED  
THE 600 HP. SINGLE ENGINED FARMAN GOLIATH  
RESULTS OF FRENCH TOURING PLANE COMPETITION

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NOVEMBER 17, 1924

# AVIATION

VOL. XVIII NO. 20

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# AVIATION

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### A Suggestion for the Winslow Bill

**T**HE House Bill for the federal regulation of civil aviation is now in process of revision. From many suggestions are being made with a view to making the measure still more effective in the general approval of all aviation interests.

One of the inherent, and therefore most serious, objections to this Bill is that it creates—in the Department of Commerce—another governmental department, and that it would add another agency to the score or more that now exist and operate in this country. The selection of the personnel required by the proposed Bureau of Civil Aviation is also a matter that causes concern. The aircraft inspectors only that the Bureau would employ adopt as a higher class for carriers in those now engaged in civilian flying.

As a substitute plan that has many supporters at the present time, it is suggested that the licensing of civilian pilots and the inspection and inspection of aircraft should be placed for the present under the Air Mail Service. This plan would save the creation of a new Government bureau. It would prevent further division of government air activities. It would dispose with the functions of a new staff of several experts, be made for the things that would avoid overlapping and confusion.

The Air Mail has stations and employees from coast to coast. It is located in a way in which everyone has access. It staff knows airplanes and pilots. Its sound, common sense methods in airplane operations have earned it the present approval of the flying community and the public. It has been equipped with the facilities that would be used by the members of civilian aircraft operations. If the Air Mail were entrusted with this work, there would be no doubt that it would come forward to pilots and aircraft on practical matters rather than to satisfy a bureaucratic interpretation of the regulations. As the Air Mail's work is limited at the best of departure and arrival of the mail, the work of inspection and licensing could be fitted into its working schedule with but a fraction of the cost that the proposed Bureau of Civil Aviation would require. At flying stations where the Air Mail has no stations, it could appoint resident aircraft inspectors from experienced local aviators; talent, and the best part of all would be their headquarters.

It will be seen from the above outline that the work of civil aircraft operations could be fitted into the Air Mail with a minimum of expense and with a maximum chance of success and general approval. Without going further into the advantages or possible drawbacks of such a plan, the suggestion is presented in our opinion. It should be added that it is made without the slightest knowledge of the Air Mail officials. There seems to be no reason, points in favor of it, to make this very necessary function of civil aviation operations over to the Air Mail that Aviation proposes it

as a solution of the controversial problems which the Winslow Bill now involves.

### The Contest Committee

**T**HE excellent work of the N. A. A. Contest Committee at Detroit, St. Louis and Boston justified the work of the important group and would make numerous announcements, were it not for the fact that a greater part of this prize is due to the pilots and persons who looked after the details. The selection of the personnel of the membership of the Contest Committee should receive due consideration so that any difficulties of the past may be overcome in the future.

The following information as to the way pilots are handled by the Aero Club of France, which has the closest connection with the F. A. I., should be carefully studied to see whether the Contest Committee is following the best practice in its procedure.

"The Comité d'Aviation de l'Aéro-Club de France which controls contests is composed of four or five members. The contest structures are represented on it, having, I believe, about fifteen members. These contesters are members of the Fédération Aéronautique des Nations de France. When a meet is to be arranged, a meeting is called of the Aviation Commission and the technical government representatives who discuss the method of procedure, rules and regulations. Over time, in France, they do not have one big contest group everything into one working. There are a number of meets from time to time. This gives a steady reference for a continued and healthy competition."

When the French system differs from ours is that in France they have representatives of the aircraft manufacturer at all meetings so that their opinions are considered before plans are made, and also that all the principal contests are not held in one place.

### Handicap Airplane Races

**I**f handicap races are to become popular these must be more widely placed around the preliminary trials. In the case of civilian planes at Bathurst every other contestant except the winner made three laps during the trials. After the winner's speed was not correctly estimated, or also changes not contemplated when the trials took place were made before the race. Just when happened is not known, but the results should make the Contest Committee of the N. A. A. extremely careful in the future of the method of handicapping. Trials should be run over the course to be used and race conditions determined as early as possible. As a further safeguard it is suggested that if a pilot exceeds his handicap trial by ten per cent or should be automatically disqualified. Handicap races will become increasingly popular if they are handled judiciously, but not if the handicapping is subject to controversy.

# The Driggs-Johnson Model DJ1 Light Plane

Efficient One-Place Monoplane Fitted with 4 Cylinder 80 cu. in. Motorcycle Engine Described

This light plane is one of the first planes of its type to be constructed in this country. It was built primarily for entry in the Dayton Daily News and Dayton Engineers' Club contest held at Dayton, Oct. 5 and 6. The Johnson Airplane and bogley Co. being in close touch with commercial aviation in general also believes that this light ship will fill a very definite need for an airplane with low start cost and low upkeep.

The designer of the DJ1, Ivan H. Driggs, is an aviator of wide experience, having been formerly assistant chief engineer with the Dayton-Wright Co. under Col. V. E. Clark

doped and vanished. The airframe are constructed of welded steel tubing, fabric covered.

Construction.—The tail surfaces are of welded steel tubing, fabric covered. They are completely locked with Rivets.

Landing Gear.—Two 400 x 50 mm. Palmer wheels are fitted. The construction is evident from the photographs. The design is of composite steel tubing, the bearing Roring type.

Flight Qualities.—No performance test has been run as yet, so that no definite figures can be given. The high speed



Then, view of the Driggs-Johnson DJ1 one-place light plane (80 cu. in. 4 cyl. motorcycle engine) which won the Dayton Daily News Trophy race

Consequently this airplane cannot be considered an amateur air attempt. All construction work except motor was carried on at the shops of the Johnson Airplane and Bogley Co. at Dayton.

## Description

Features.—The fuselage structure is of welded steel tubing, covered with fabric. All of the cockpit, the floor and the wing are at top location. The cockpit is completely covered with a perfect streamlining, fitting into the fuselage of the fuselage. The rear view gives a very clear idea of that construction.

Engine.—The engine used is a four cylinder 80 cu. in. air-cooled motorcycle motor of standard make. Thrust bearings and propeller hub extension were built into the motor in place of the regular flywheel housing. Cooling is accomplished by means of a scoop at the front under the motor, which forces air under the oil pan and then up the left side of the engine into the exhaust and intake valve chests. The air is then drawn across the engine from the left to right by means of a scoop on the right hand side of the wing. An air scoop is also provided at the front of cylinder No. 1 and is in the circulation. Likewise, a small lower scoop is at the top of each cylinder. The photographs give some idea of the cooling arrangement. The fuel tank is mounted in the leading edge of the wing, directly over the engine, providing an excellent fuel gravity to the engine.

Wing.—The wing used is DRA-35. The wing is entirely internally braced, with 1/32 in. thick plywood covering for internal stiffening. The spars are laminated spruce and laminated. The spar web is graded into and from the root as the stresses demand by leaving off unnecessary laminations. The ribs are 1/32 in. spruce rather than ply in web, and 1/32 in. 2/56 in. spruce battens raised and glued to each side. All of the rear spar ribs are covered with surgical tape.

is somewhat in excess of 80 mi./hr., no determined approximately by means of a level line. The engine is a very recent and although it is very light, the fuel is perfectly satisfactory. The airplane then very much more powerfully, and does not give the pilot any feeling of air current such as is often, is present in small airplanes.

During the race on Oct. 4, because of poor start, three first place prizes were made in very rough and difficult country. The pilot, J. M. Johnson, was forced to land the first time in a newly plowed field, which he negotiated without mishap in spite of the fact that he was not in the air. The second time he was forced to land in a field of corn, which he negotiated without mishap. After getting off it was necessary for him to back up around a number of trees before getting clear. In a word the DJ1 is exceptionally easy and comfortable to fly. It keeps it in work better than a JNA and absolutely no fuel is needed on the tank. A slight motion of the finger or foot is sufficient for control on the most severe weather.

## Specifications

Wing area	10 sq. ft.
Wing span	10 ft.
Wing chord at root	10 ft.
Wing chord at tip	10 ft.
Wing thickness	10 ft.
Wing weight	10 lb.
Wing load	10 lb./sq. ft.
Wing strength	10 lb./sq. ft.
Wing material	10 lb./sq. ft.
Wing construction	10 lb./sq. ft.
Wing design	10 lb./sq. ft.
Wing test	10 lb./sq. ft.
Wing results	10 lb./sq. ft.
Wing conclusion	10 lb./sq. ft.

## Weight Schedule

Wing weight	10 lb.
Wing material	10 lb.
Wing construction	10 lb.
Wing design	10 lb.
Wing test	10 lb.
Wing results	10 lb.
Wing conclusion	10 lb.

Wing area	10 sq. ft.	10 sq. ft.
Wing span	10 ft.	10 ft.
Wing chord at root	10 ft.	10 ft.
Wing chord at tip	10 ft.	10 ft.
Wing thickness	10 ft.	10 ft.
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Wing results	10 lb./sq. ft.	10 lb./sq. ft.
Wing conclusion	10 lb./sq. ft.	10 lb./sq. ft.

## French Touring Plane Cross Country Race

France encourages the development of engine airplanes by a series of practical test and cross country flights, the one of which gets substantial prize money. This year there have been held at France cross country flights for light planes, for medium transport planes, for medium commercial planes and finally for tractor planes, so we should call it, sport planes. The last contest on test was managed by the Aero Club of France under the auspices of the French Section for International Aviation.

The winner was 1,200 m. in length and consisted of a number of different flights. Three days were allowed to complete the circuit, and a maximum speed of 45 mi./hr. was to be maintained. The winner was a small, light, secondary airplane being put on speed and the number of passengers were not to exceed two. The winner was a small, light, secondary airplane being put on speed and the number of passengers were not to exceed two. The winner was a small, light, secondary airplane being put on speed and the number of passengers were not to exceed two.

Twenty-five planes were entered, but none had not received the proper papers from the French Section and so were eliminated. Of the sixteen planes which entered only six did not arrive at the finish, the elimination being due to a lack of fuel or lack of engine power. The winner of the whole circuit was a small, light, secondary airplane being put on speed and the number of passengers were not to exceed two. The winner was a small, light, secondary airplane being put on speed and the number of passengers were not to exceed two.



The winner of the DRA-35 fitted with an inverted type Liberty-12 engine. The success of this engine has prompted its adoption in the Army and Navy Air Services

First place was won by a two-place Potez type VIII (50 hp. Anzani), piloted by Labrousse at a speed of 45 mi./hr. Second place went to a three-place Morane-Saulnier type 30 (120 hp. Gnome) piloted by Frenet at a speed of 42 mi./hr. Third place went to a three-place Gourdou type G12 (120 hp. Gnome) piloted by Buis at a speed of 40 mi./hr. There were five Potez planes entered, all fitted with Anzani engines and all five were present at the finish. Miss Boland, the only woman entrant, finished fifth. She flew a two-place Gourdou type G12 fitted with an 80 hp. Elgin motor.

The Libbey (P.H.A.) amphibian fitted with an 80 hp. Elgin motor finished seventh. This machine is a very close looking thing but of the power type. The wheels are attached to the lower end of a V-shaped bracket, the upper ends of which are attached at points approximating the wing hinges. There is no shock absorbing apparatus, but very large, easily inflated tires are used. The wheels when hit come for landing flares in so an axle which runs through the hull.

The only novelty in the motor line was the new Gnome A.C. Heliote the Libbey Co. combined this to make a motor engine. The A.C. is an air cooled motor engine which seems to have functioned well in the two machines to which it was fitted.

## Inverted Aircraft Engines

Editor, Aviation—

I note on page 317 of your issue of Sept. 1 reference to a claim for having made the first experiments with inverted aircraft engines. May I point out that in 1913 the French Gignoux firm built and sold a 40 hp. Clark inverted engine for airplanes which was used by several French pilots with fair success? Further, Mr. Hildebrand of the Hildebrand firm in 1914 turned his 6-cyl. J.A.P. inverted engine upside down in the single engine Morane-Saulnier motor-plane which was flown by the late D. G. Gignoux. This engine ran quite satisfactorily upside down, that it had ever been put upside down is an undisputed point.

We will see that experiments with upside down engines in Europe are of a considerable antiquity. Except in the case of some of the older inverted engines which depended very largely on an excessive oil consumption to maintain them in a reasonable temperature, experience showed that trouble due to oil got up into it was a serious one. In fact, inverted engines and wing wings are an extremely good condition, the trouble very rarely arises. The inverted engine has not been used in this country in the way many believe that inverted engines of that sort matched upside down would be extremely successful for inspection and overhaul.

W. H. Searns  
Technical Editor, The Aeroplane  
London, Sept. 15, 1928

# The 600 hp. Single Engined Farman Goliath

New 600 hp. Farman Engine Replaces Twin Engine Arrangement  
In the Goliath Air Liner

When, toward the end of 1919, civil air transportation had its modest beginnings in the passenger and freight services instituted between Paris and London by French and British air lines, all but one of the types of planes used were one-engine aircraft. The sole exception was the Farman Goliath 12-passenger transport, which, though originally built down as a bomber, had been turned into a passenger carrier while it was still in the design stage. Then the Goliath was actually the first public air transport that was specifically designed for the requirements of passenger and freight carry-

which was held that year, comprising three 600 hp. engines without a landing gear in 36 hr. at an average speed of 41 mi./hr., including all stops. The useful load carried on the journey was 22 tons, and the gross weight 41 tons. The high speed of the machine, at 6000 ft., was 360 mi./hr. The four-engined, so-called "one-engine," Goliath was produced for the 1125 Air Transport Grand Prize of the French Air Department. The ship was equipped with 270 hp. Hispano engines, which were mounted on tandem axis, in bay nacelles, and which actuated tandem propellers. Although



The Farman "Goliath" 12-passenger transport (two 300 hp. Renault engines) used on the Paris-London line.

ing. The night bomber that was derived from it, and with which the French air force is now equipped, only appeared a few years later. It is difficult to see an original type design power—as numerous aircraft designers contend—without soundness of design, then the Goliath must rank high among such accomplishments. For though a number of minor changes have been incorporated in this ship, less the experience of the years' extended operation on the northern routes of France (in England, Belgium and Holland), basically the type has remained unaltered in constructional principles as well as in looks. A 1924 Goliath looks exactly like a 1919 Goliath, except for the coating of the power plant which actually varies with the type of engine employed.

## The Multi Engined Types

Originally, the power plant of the Goliath consisted of two 230 hp. Hispano engines, driving tractor propellers. Later, in order to increase the speed of the ship—which was then about 90 mi./hr.—two 300 hp. Renault 12-cyl. Vee type engines were substituted, and since of these ships were also fitted with 170 hp. Lorraine-Dietrich 12-cyl. Vee engines for emergency purposes. Although the two-engined Goliath has remained the standard passenger-carrier on the French cross-Channel services, experimental three and four engined ships of this type have also been produced by the Farman factory.

In 1923 there appeared the three-engined type, fitted with 300 hp. Hispano engines, the third engine being mounted in the nose of the fuselage. The ship was—and was the only contestant in French—the French Air Transport Competition

accommodation had been provided for eighteen passengers in addition to a crew of three, the plane was only enabled to set a pay load of eight passengers and baggage owing to its large amount of "overweight" navigation and safety equipment imposed by the French Air Department for this competition. The ship was second prize, the winner being the Fokker 12-cyl. engineless competitor.

The "one-engine" Goliath has a wing area of 1509 sq. ft. and weighs 2084 lb. empty, and 35,000 lb. loaded. In its above mentioned competition the ship covered three times normal speed, maintaining 1280 mi. in length, at a normal speed of 70.6 mi./hr.

The latest alteration in the power plant of the Goliath results in the suppression of the wing engines, and the substitution thereof of a single 600 hp. Farman multi 18-cyl. engine in the nose of the fuselage, with a single tractor propeller. As may be seen from the accompanying illustration, the single-engined Goliath does not materially differ in look from its multi-engined ancestor. The suppression of the wing engines has entirely "cleared up" the wing structure, but otherwise the latter is except for minor variations in dimensions, identical with that of the two-engined Goliath.

## The 600 hp. Farman Engine

The Farman 18-cyl. engine has a normal output of 600 hp. at 1600 rpm. The overrunning power, developed at 1690 rpm, is 720 hp. The engine is a 18 cylinder V type, with three banks of six cylinders set at an angle of 45 deg. The bore is 170 mm., the stroke is 200 mm. and the cylinder capacity 473 liters. At its normal rating the piston speed is 575 ft./min. and the oil consumption 19 g./A.

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propeller rotation gear of 2:1 ratio is specially fitted, but this can be replaced by reduction gears giving 1:84, 1:137, 1 and 1:13-1 ratios, or by a direct drive. The direct drive used with dry, without oil, but with all accessories, 1718 ft. The 2:1 reduction gear used weighs 118 lb. more. It then follows the weight of the electric starter is not included; this weight, together with the starting battery, which can also be replaced for lighting the cabin, 115 lb.

## Construction Details of Engine

Following are the main constructional features of the 600 hp. Farman engine. The cylinders are cast of steel in pairs and secured together. They are enclosed in a steel water jacket which is soldered into the cylinder blocks. The top of each cylinder is ground to remove the cylinder head, which causes the four valves seats and plug threads.

The crankshaft is of the overcrank type, the cranks being at a 90 deg. at 180 deg. The connecting rods consist of a main rod of 1 1/2 inches and of two rods of tubular section which are articulated on the top end of the main rod. The pistons are aluminum alloy castings.



The new, single-engined Farman Goliath air transport. The close-up on the right side shows the mounting of the 600 hp. Farman 18-cyl. engine and the Lanchester radiator.

The valve gear comprises two camshafts which are located inside the upper part of the crankcase. One camshaft controls the valves of one cylinder row, and the other camshaft the valves of the other two rows. The valves are actuated by means of rocker arms and pushrods. Double ignition is provided in terms of two magneto and Delco ignition systems. Lubrication is effected by means of a gear pump which draws the oil from the tank and sends it to the crankshaft bearings from where it is projected onto the main rods and into the cylinders. The oil returning to the crankcase is forced back by two other gear pumps into the oil tank. A hand pump draws the oil and delivered into an oil radiator where it is cooled for further use.

Exhausting is by Scotch combustion fitted with triple D-fitters. There are two single and two double carburetors, each of them feeding three cylinders. The gasoline system is of the pressure type.

The water circulation system is of the centrifugal type, with one pump.

A 30-watt electric starter of 24 volts, operating from a 120 v. a.c. generator, forms the normal equipment of the ship.

## Engine Compartment and Cabin

One of the accompanying illustrations shows the main point in which the big engine is mounted on the new Goliath. It will appear that their ground sheet takes away the support to the fuselage, and that the whole engine mounting can be taken down by unscrewing a few nuts, and removing the surrounding bolts. The excellent accessibility of the engine compartment is a noteworthy point.

The cooling of the engine is effected by two Lanchester radiators which are suspended from the fuselage just ahead of the lower wing.

Owing to the mounting of the big engine in the nose of the fuselage, the internal arrangement of the cabin has been simplified. The control cockpit, seating pilot and mechanic, is situated right behind the engine in such manner that the mechanic has direct access to the engine and can carry out minor adjustments in flight. A radio transmitting and receiving apparatus is contained in a sound proof compartment at the side of the mechanic, so the latter can act as a radio operator when he is not otherwise employed. The fuel supply, sufficient for 3 h. flight, is stored in seven placed behind the control cockpit, where a double fire proof bulk head separates them from the passenger cabin. The latter accommodates eight passengers in comfortable armchairs, but two more can be turned into folding seats. A lavatory and a baggage hold are fitted at the end of the cabin.

The landing gear is perhaps the most striking deviation from Farman practice for this classed type of ship. The sole-less "streamer legs" cross-braced by wires, which were in

characteristic of the previous types of Goliath, have been replaced by two modern Vees which are joined by a streamer fixed cross-member. The two wheels are mounted on a divided axle, articulated in two points, and rubber springs.

The fuselage, the wings and the tail end are, with the exception of the fittings, entirely constructed of wood. Sheet aluminum is used for the covering of the engine compartment and the passenger cockpit, but all of the latter the tailplane is fabric covered, as are the wings and the surfaces.

## Performance Figures

Following are the characteristics of the 600 hp. single engined Goliath:

CHARACTERISTICS OF THE 600 HP. FARMAN ENGINES (Lanchester)			
Power	480 h.p.	Weight empty	2,600 lb.
Weight	11,800 lb.	Power 121	1014 h.p.
Stroke	17.5 in.	Power 121	1014 h.p.
Speed	120 mi./hr.	Power 121	1014 h.p.
Stroke	17.5 in.	Power 121	1014 h.p.
Speed	120 mi./hr.	Power 121	1014 h.p.

In flying trials, the ship being loaded with 4400 lb. of fuel load at speed 3200 ft. given in the above table, the high speed attained was 121 mi./hr., the maximum flying speed 100 mi./hr., and the take-off and landing speeds both 50 mi./hr. With the load the ship attained 20,500 ft. in 21 sec. The remarkable weight carrying feature of the 600 hp. Goliath are furthermore demonstrated by the fact that this ship holds the world altitude/weight records for 1926, 1928, 1930 and 1932 for the 1000 lb. record made May 6, 1933 by Loren Cooper, in 18,800 ft., the record for both the 1000 and 2000 lb. records, made by Loren Cooper on May 8.





developed engine is the slightly larger ones. And construction of both the offset plane and engine becomes more involved and difficult as the size is increased. Whatever the reason for the unsatisfactory engine situation, the demand for engine engines is certain. The Editor of *The Aeroplane* says in the Oct. 6 issue, "It has been very generally held that the primary fault with the motor was the setting of the limit of engine speed as low as 1,100 r.p.m. But one in five firms believing in the value of the 1,100 or even the 1,500 r.p.m. engine. The essential feature of the practical light aeroplane is that it should be small, light and cheap, and require only a moderate engine power to fly it. Provided it complies with these requisites, it really does not matter whether it has an engine of 400 c.u. or of 4 h.p. Firstly, one is fairly firmly convinced that if the light aeroplane is to be of any use at all it must be fitted with an engine having a large head stroke, a low compression ratio, and running at a low speed. There is no reason why such an engine should not be built for very little more weight, and probably for less cost, than the more modern type. Secondly, for a two-seater one would choose it if it were possible a 2,000 c.u. engine developing about 40 h.p. at 1700 or 1800 r.p.m."

#### What Is a Light Plane?

The Editor of *Flight* remarks on the same view, "Into the question of future engine capacity we do not wish to enter at the moment beyond pointing out that two views appear to be open. One is to go on developing the 1,100 r.p.m. engine until reliability combined with low weight and high power is attained. Another is to go for the larger engines, say, 2,000 c.u. and to limit the piston speed. The problem is very complicated, and very good arguments can be put up for either course."

The attitude of the French is well known. Last year they raised the limit of light plane engines to 1,500 c.u. but designers and 2,000 c.u. for single-seaters. They are developing a different type of product, the sport plane, very useful and desirable in its field, but burdening on the line of those machines which are no longer for sport planes.

All this seems to indicate a lack of consensus as to the objective of light plane development. What is the difference between a Caudron 25 HD with its 80 h.p., carrying two passengers and a single-seater modified light plane with a 50 h.p. motor? What does one see in the new Leard Commercial

or the next little Swallow with their four seats and the 100 h.p. GEXA motor? The Swallow is more of a light plane than most of the French single-seaters. Where are we going with our light planes? Is the inevitable end to be a merging of the types and a loss of identity of the light plane as such?

#### The Official Results

The following are the official figures of the performance obtained by the non-eliminated competitors in the course of the trials.

**Round-robin: No. 7. No. 4. Pilot: M. W. Parry. High speed 71.11 m.p.h. Low speed 59.05 m.p.h. Speed range 16.77 per cent. Get-off 255 yd. Pull-up 324 yd. Boom from 15 to 54 sec. Miles from 737.6. 1st Air Mailer price £2,000.**

**Round-robin: No. 3. Pilot: C. F. Evans. High speed 65.16 m.p.h. Low speed 59.73. Speed range 68.33 per cent. Get-off 215 yd. Pull-up 309.6 yd. Boom from 16 to 35 sec. Miles from 512.5. 2nd Air Mailer price £1,000. State of Dunbar's price £500 for Get-off and Pull-up.**

**Round-robin: No. 35. Pilot: F. P. Barnham. Low speed 37.43 m.p.h. Get-off 255 yd. Pull-up 73.05 yd. Boom from 18 to 25 sec. Capt. C. B. Wilson price £150 for Get-off and Pull-up.**

**Crossed: C.A.A. No. 3. Pilot: Ft. Lieutenant. Chaper. Boom from 17 to 55 sec. Miles from 762.5. S.M.H.T. price £100 and R.C.M.C.T. price £150 for maximum distance flown.**

**Round-robin: No. 14. Pilot: Sq. Leader Langton. Low speed 31.42 m.p.h. Get-off 230 yd. Pull-up 12.80 yd. Boom from 8 to 24 sec. Miles from 490.**

**Round-robin: No. 18. Pilot: Sq. Ldr. Morda Douglas. Low speed 37.22 m.p.h. Get-off 201 yd. Pull-up 70 yd. Boom from 16 to 4 sec. Miles from 100.**

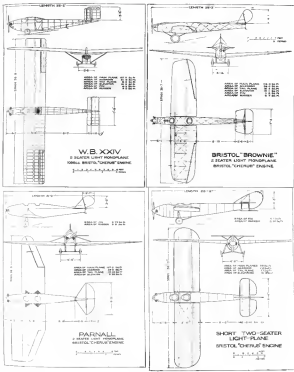
**Without Road Power: No. 5. Pilot: S. H. Gaskett. Boom from 2 to 31 sec. Miles from 125.**

We are indebted to *Flight* for the outline drawings of some of the new French light planes that appear on the opposite page. Another group, illustrating light light planes will be published in our next issue.

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#### Specifications of the Light Planes Entered in the Lympne Meet

Entry	Plane	Engine	Power	Speed	Altitude	Endurance	Price
1	British Aeroplane Co. Ltd.	Hispano-Suiza	100 h.p.	100 m.p.h.	10,000 ft.	1 hour	£1,000
2	British Aeroplane Co. Ltd.	Hispano-Suiza	100 h.p.	100 m.p.h.	10,000 ft.	1 hour	£1,000
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99	British Aeroplane Co. Ltd.	Hispano-Suiza	100 h.p.	100 m.p.h.	10,000 ft.	1 hour	£1,000
100	British Aeroplane Co. Ltd.	Hispano-Suiza	100 h.p.	100 m.p.h.	10,000 ft.	1 hour	£1,000











## Publisher's News Letter

The election of President Coolidge will have a very direct influence on the future of our national aviation policy. Probably the full significance of the election has not been realized by those who have the interest of aviation at heart. The possibilities of what might have happened can be considered without losing any perspective influence on the picture. A change at this time would have created uncertainties that would have put off consideration of the most important problems for a year at least as the new Congress does not assemble until December, 1925. Then, too, the tendencies of Senator La Follette and of Mr. Davis in the direction of a strong air force are not known, so that from a purely negative point of view the continuation of the present administration will at least stabilize the situation.

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From a positive view there can be nothing but congratulatory emotions. That the business conditions will become better has been indicated immediately. With the certainty of a known program for four years, business will now be able to make its plans for the next few years. This will have a very helpful influence on the aeronautical industry for it will mean that capital will feel free to consider this field as an investment proposition. With business in a throng condition, the tendency toward progress in the field of commercial aviation is inevitable. New uses for aircraft will be tried out that would not be undertaken if there were the possibility of an economic depression on the horizon. Viewed entirely from the angle of advancing commercial aviation, the election holds a promise of prosperity that cannot be denied.

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By far the greater stimulus to aviation will come in the governmental field. President Coolidge has stated an unshakable tenet in his important interest in the aircraft problem of the service. He wants to know whether the battleship is still as much a factor in naval warfare as it was before the development of aircraft. He, also, doubts will want to know whether we are getting the maximum results from the expenditures we are making for our naval defense. Out of all this aviating interest shown by the President there must come some definite policy that will give a progressive development that has been lacking in the past. For these reasons the election must cheer

everyone who has even the slightest knowledge of the critical condition in which aviation in the country finds itself at present.

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The *New York World* has followed the election period of crowded news columns by a capsule of our air services that will make aviation history. Mr. Ralph Pollock was at the Dayton meet and evidently what he saw convinced him that something was radically wrong with our governmental aeronautical progress. The series which *AVIATION* hopes to repeat in full gives substantial confirmation of all the criticism that has been made about excessive costs, duplication, uncertainty of figures, and a general aimless wandering by the service. As these articles will appear throughout the country in other papers the effect will be strongly toward a change in policy or better still, will bring about some definite national policy.

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At the same time there will appear a series of articles in the *Herald* papers by Charles Levy, Editor of *The Aeroplane*. These articles will deal with American aeronautical problems from an angle in popular style. Mr. Grey has also sent *AVIATION* a series of articles on American Aviation that will be of great interest to our readers. Not only do they come from the pen of a writer whose style is incisive and brilliant, but also from one who has taken such a personal interest in American aeronautical progress that he is better informed on many particulars than those who are so close to our problems that they cannot view the situation as its entirety. As Mr. Grey is also the Editor of "All the World's Aircraft" he has a world wide viewpoint that gives his writings the range of a view seen through a very wide angle.

Frank, open discussion is needed at this time. With *The World* and its allied papers, and the *Herald* group under the editorial direction of Arthur Brisbane giving the whole problem of aviation their best editorial attention, the time seems ripe for a forward movement by the aeronautical front line. Hereafter aviation has seemed to be on the defensive. Now it is not only undertaking an offensive but by cold facts has placed the obstructionists on the defensive.—L.D.G.

## A Suggested National Air Policy

That a National Aviation Policy is needed by the United States is obvious. To get such a policy in concrete form *AVIATION* requested several thoughtful friends of aeronautical progress to make suggestive and constructive recommendations. Some of them are given below and will be printed each week with additions, omissions and such other changes as appear to be helpful toward the formulation of a sound national air policy. Readers of *AVIATION* and others can render no greater service to the cause of aeronautical progress than contributing their comments and suggestions.

### GOVERNMENTAL.

- A continuing program of aircraft development both governmental and commercial.
- A civilian, charged with recommending a national air policy, is needed in the Government.
- \*Cabinet Aircraft committees in the House and Senate to hold aircraft hearings where civilians as well as government officials can express their opinions. \*Composed of five.
- A detailed aircraft budget for all Governmental Departments, and an annual statement of all expenditures.
- An experienced staff of flying officers at the head of all governmental air defense services.
- Coordination of all procurement and experimental aircraft work of the government under one agency.
- \*Coordination of the aircraft capitalization development of the government having procurement in the various branches themselves.
- Limitation of government manufacture to repair of aircraft and specialized work that cannot be done by private firms.
- \*No limitation on experimental construction.
- The elimination of the duplication of aerial functions and facilities by government departments.
- A country wide Air Mail system of trunk lines connecting the principal cities of the country.
- \*Refinement law for air mail pilots.
- Establishment of a National Airways System through cooperation of the Federal Government with States and Cities.
- \*A landing field in every large city.
- A national aircraft law that will regulate creation, administered by practical pilots and experienced aeronautical engineers.
- \*and federal air police.
- Membership of the United States in the International Convention for Air Navigation.
- \*Increased governmental expenditures for aerial development.
- \*Encouragement of aviation rather than subsidy.

### COMMERCIAL AIRCRAFT OPERATION.

- Creation of commercial air lines by private enterprise or government subsidy.
- Encouragement of participation by private companies in aircraft races and competitions.
- Encouragement of the training of pilots by civilian schools.
- Creating an Expert du Corps among flying men all over the country by frequent gatherings at aviation meets.
- \*Encouragement of safe and sane flying.
- \*A continuing organization, including representatives of all important points of view in aeronautics, for the discussion of standards for aeronautics where standardization is desirable.

### INDUSTRIAL AIRCRAFT CONSTRUCTION.

- Recognition that a sound aeronautical industry is a prime necessity of our National Defense.
- An active industrial association that will coordinate the aircraft industry and defend it from attack.
- Encouragement of the designing of new types of aircraft by manufacturers by allowing them to obtain their proprietary rights.
- Construction of manufacturing firms on specialized types of army and navy aircraft.
- \*When production demands are heavy.
- Encouragement of research by construction, universities and other agencies as well as by the government.
- Encouragement of an annual design competition for commercial aircraft.

### CIVILIAN.

- A national aeronautical organization composed of public spirited citizens that will take a strong position of leadership in national aeronautical policy.
- \*Unification of all aeronautical organizations into one national association with chapters in all cities and towns.
- An Annual Aviation Week during which the country will think of aerial progress.
- \*32 such weeks.
- The formation of local aero clubs by them for the purpose of stimulating flying in all localities.
- Encouraging the public to fly and patronize the air mail and transport facilities.
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